



OMERE space radiation environment and effects tool: new developments and new interface

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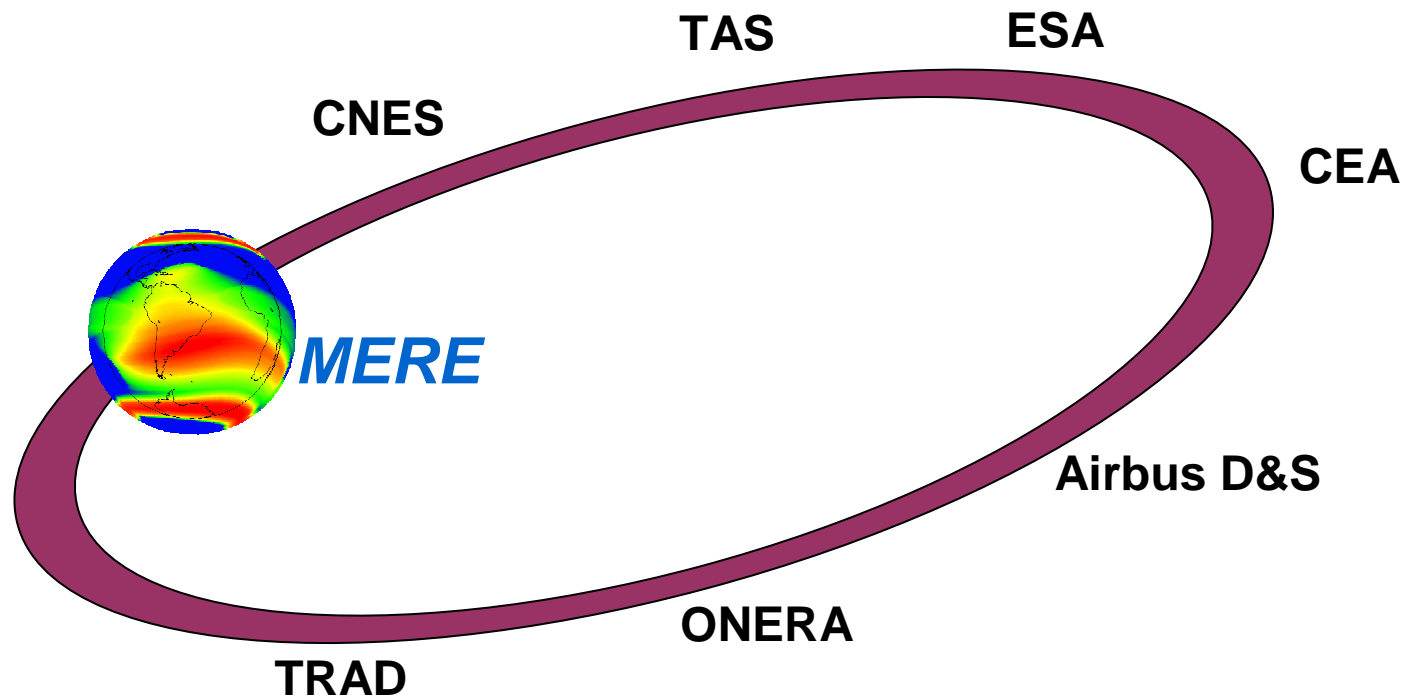
R. Ecoffet, G. Rolland and D. Standarovski, CNES

OMERE = Outil de Modélisation de l'Environnement
Radiatif Externe

■ The project

- ▶ Since 1999.
- ▶ TRAD development with CNES support.
- ▶ Freeware for space radiation environment and effects on electronic components.
- ▶ Stand alone software (no internet connection needed).
- ▶ Conceived to meet industrial requirements.
- ▶ Integrates ONERA models.
- ▶ Integrates outcomes of Research and Technology projects financed by the CNES.
- ▶ Coupling with FASTRAD®.

- The partnership



- Existing modules

- ▶ Orbit and mission definition
 - Orbit parameters or trajectory file
- ▶ Radiation environment definition
 - Radiation belt models
 - Solar proton and solar ion models
 - Cosmic Ray models
- ▶ Ionising dose
 - Dose depth curve behind Al equivalent shielding
 - Dose rate calculation along the orbit
- ▶ Non-Ionising dose
 - Using NIEL curves from the ONERA NEMO (NIEL Evaluation Model of ONERA) code.
 - Electron, proton and neutron equivalent fluence.

- Existing modules

- ▶ LET spectrum
 - Behind fixed aluminium equivalent shielding or using a sector file.
- ▶ Particle Transport
 - Electron, proton, ion behind fixed aluminium equivalent shielding or using a sector file.
- ▶ Single Event Effects
 - Component database
 - Weibull fit of ion and proton cross-section curves
 - PROFIT and SIMPA methods for predicting proton cross-section curve from the ion one
 - Mission average and along the orbit single event rate behind fixed aluminium equivalent shielding or using a sector file.

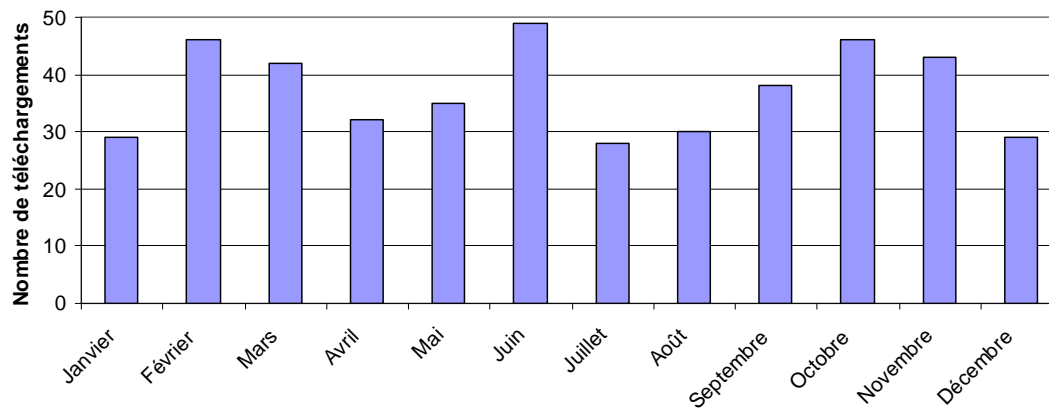
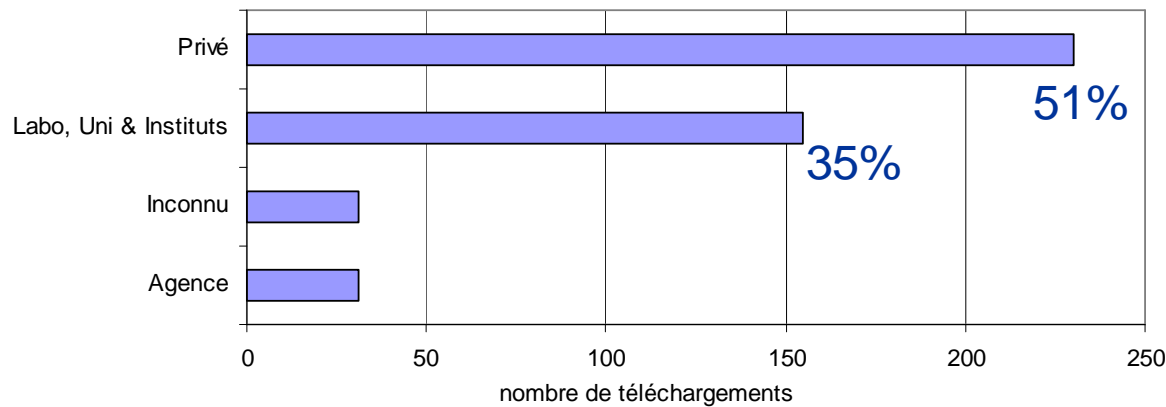
- Existing modules
 - ▶ Equivalent LET
 - LET variation inside the sensitive volume for an ion of energy E.
 - ▶ Solar Cells
 - ▶ Multi-mission calculations
 - Batch calculations of environment and effects for multiple missions.
 - Post processing tools.

- The users

- ▶ System engineering (can be a client requirement)
- ▶ Electronic component engineering
- ▶ Equipment and scientific instrument conception
- ▶ Research and development
- ▶ Education

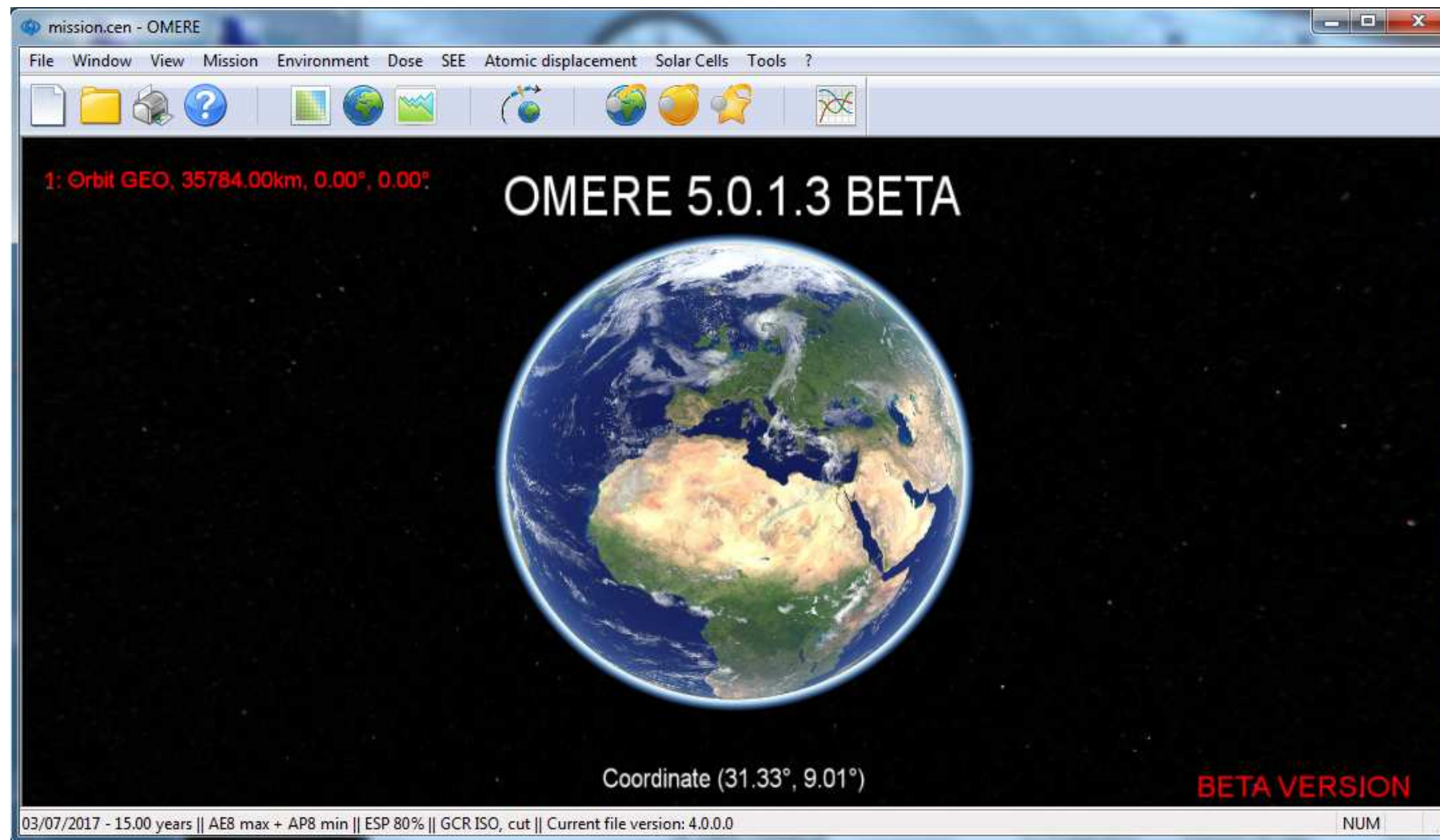
■ User statistics

- ▶ 450 downloads in 2016
- ▶ From 40 different countries



On average 1 new download/day

- Since 2015, important efforts to improve interface and optimize code
- Calculations are faster – approximate factor of x2 less
- The first version reflecting this work is v5.0, soon to be released on the TRAD web page (<http://www.trad.fr/OMERE-Software.html>)



The screenshot shows the OMERE 5.0 software interface. The main window displays mission parameters for 'Orbit GEO' and a table of orbit details. The dialog box allows setting the launch date, time, and orbit parameters.

Mission definition dialog box:

- Mission launch date:
 - Date of launch: 03/07/2017
 - Time of launch: 02:01:25
 - Solar Cycle: [button]
- Orbit(s):

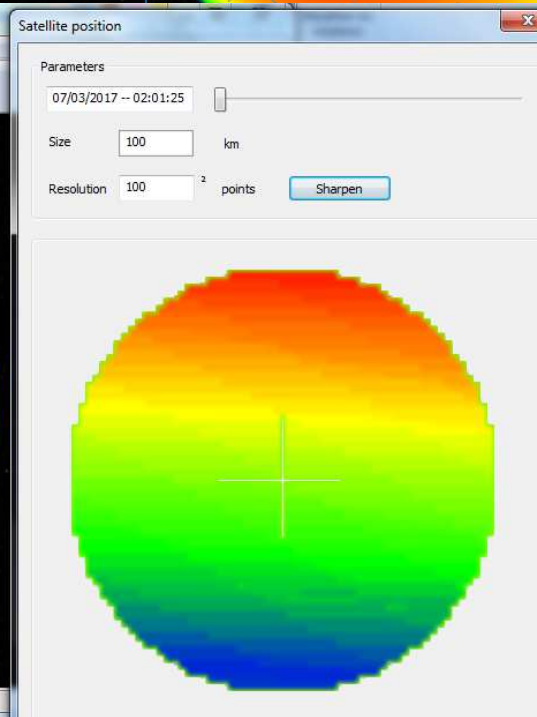
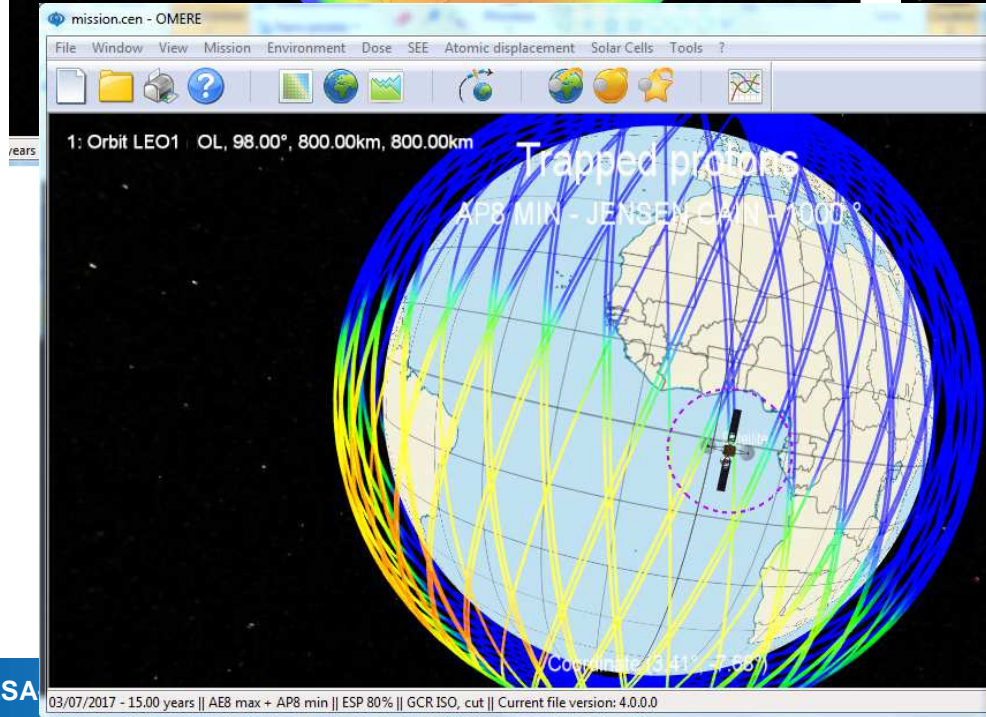
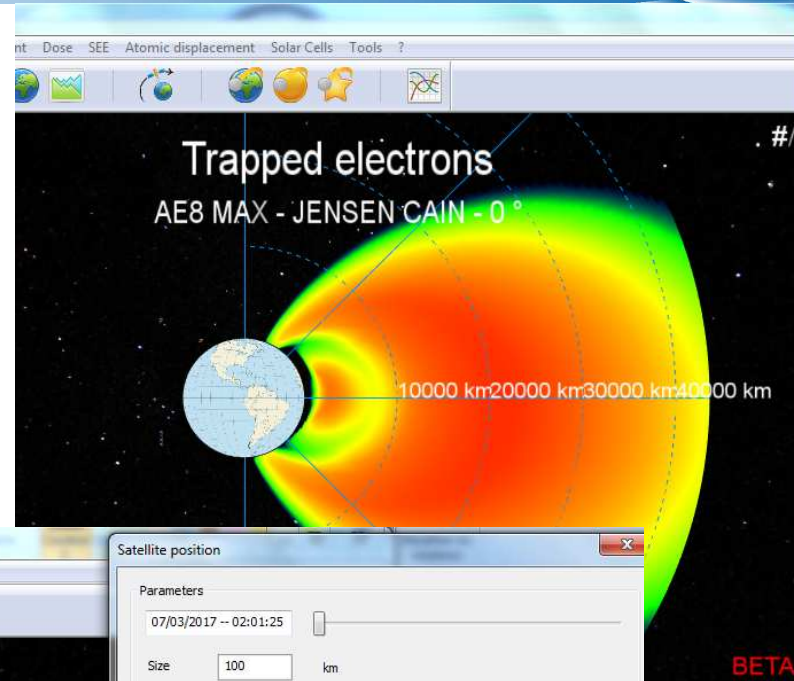
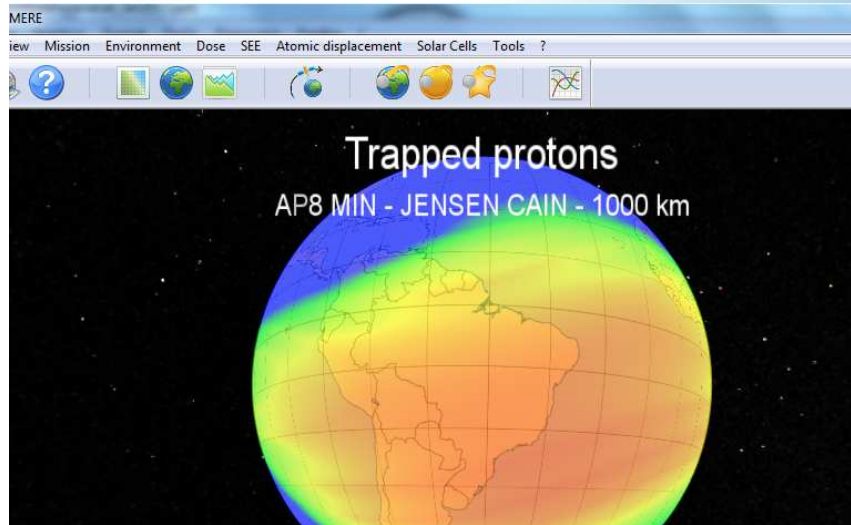
Orbit name	Inclination	Apogee	Perigee
Orbit GEO	0.0°	35784.0km	35784.0km
- Output File: C:\OMERE\orbit.dat

Main window display:

1: Orbit GEO, 35784.00km, 0.00°, 0.00°

OMERE

03/07/2017 - 15.00 years || AE8 max + AP8 min || ESP 80% || GCR ISO, cut || Current file ver



mission.cen - OMERE

File Window View Mission Environment Dose SEE Atomic displacement Solar Cells Tools ?

1: Orbit GEO, 35784.00km, 0.00°, 0.00°

SINGLE EVENT UPSET DATA

Part Type : HM0000000B Function : 256 M-bit SDRAM
 Manufacturer : HITACHI Capacity (cells) : 450000
 Technology : CMOS

HEAVY IONS TEST DATA **PROTONS TEST DATA**

Reference Report : SEE/001 Reference Report : SEE/002
 LET threshold = 4.0 MeV.cm²/mg E threshold = 23.0 MeV
 Device Cross Section = 9.91e-002 cm²/device Device Cross Section = 2.55e-008 cm²/device
 W = 19.84 W = 0.28
 S = 1.53 S = 0.23

SEU RATES - Galactic Cosmic Rays @ 1.0 u.cm-2 **SEU RATES - Trapped Protons @ 1.0 u.cm-2**

Mean Rate = 2.3e-002 seu/device.day Mean Rate = 1.8e-001 seu/device.day
 Max Rate @ 90° = 6.3e-002 seu/device.day Max Rate @ SAA = 9.8e+000 seu/device.day

GEOGRAPHICAL DISTRIBUTION OF INSTANTANEOUS SEU RATES > MEAN SEU RATE

Figure 1:

Mean SEU Rates	SEU / device.day
Cosmic Rays	2.3e-002
Trapped Protons	1.8e-001
Total	2.1e-001

Figure 2:

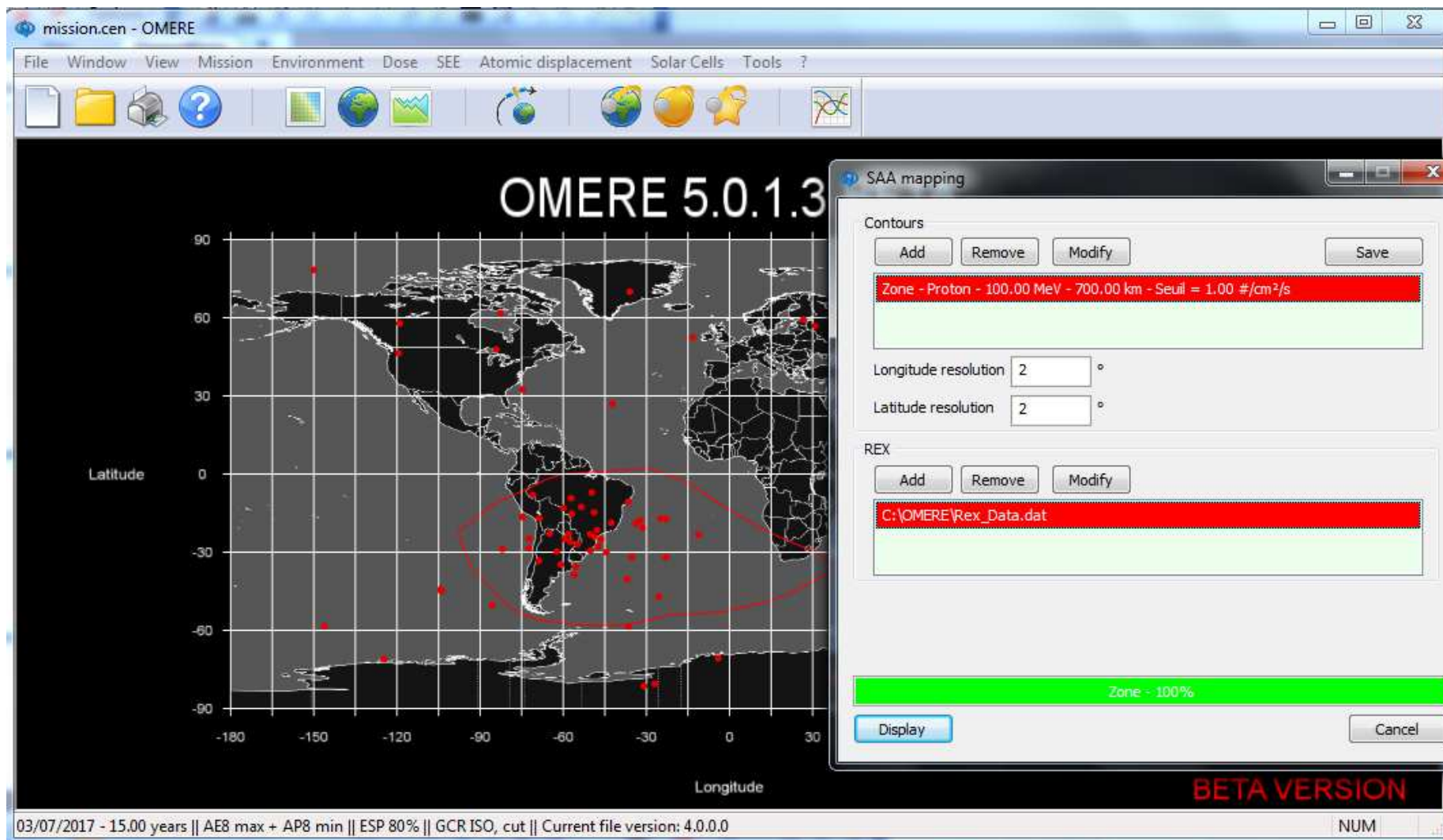
# SEU /	device.12.5 years
Galactic Cosmic Rays	1.1e+002 11.29%
SAA Trapped Protons	8.3e+002 88.71%
Total	9.4e+002

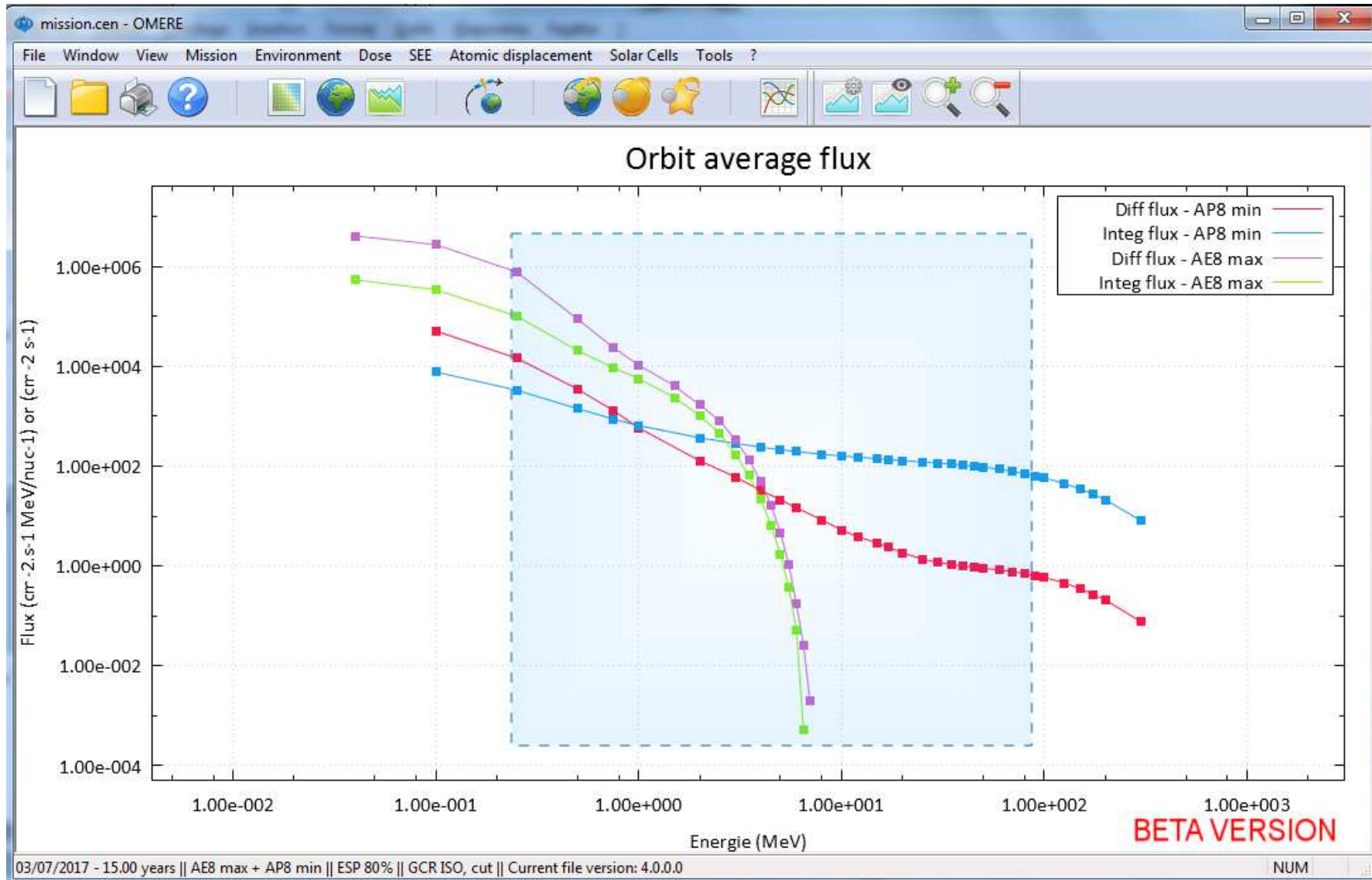
Orbit Characteristics

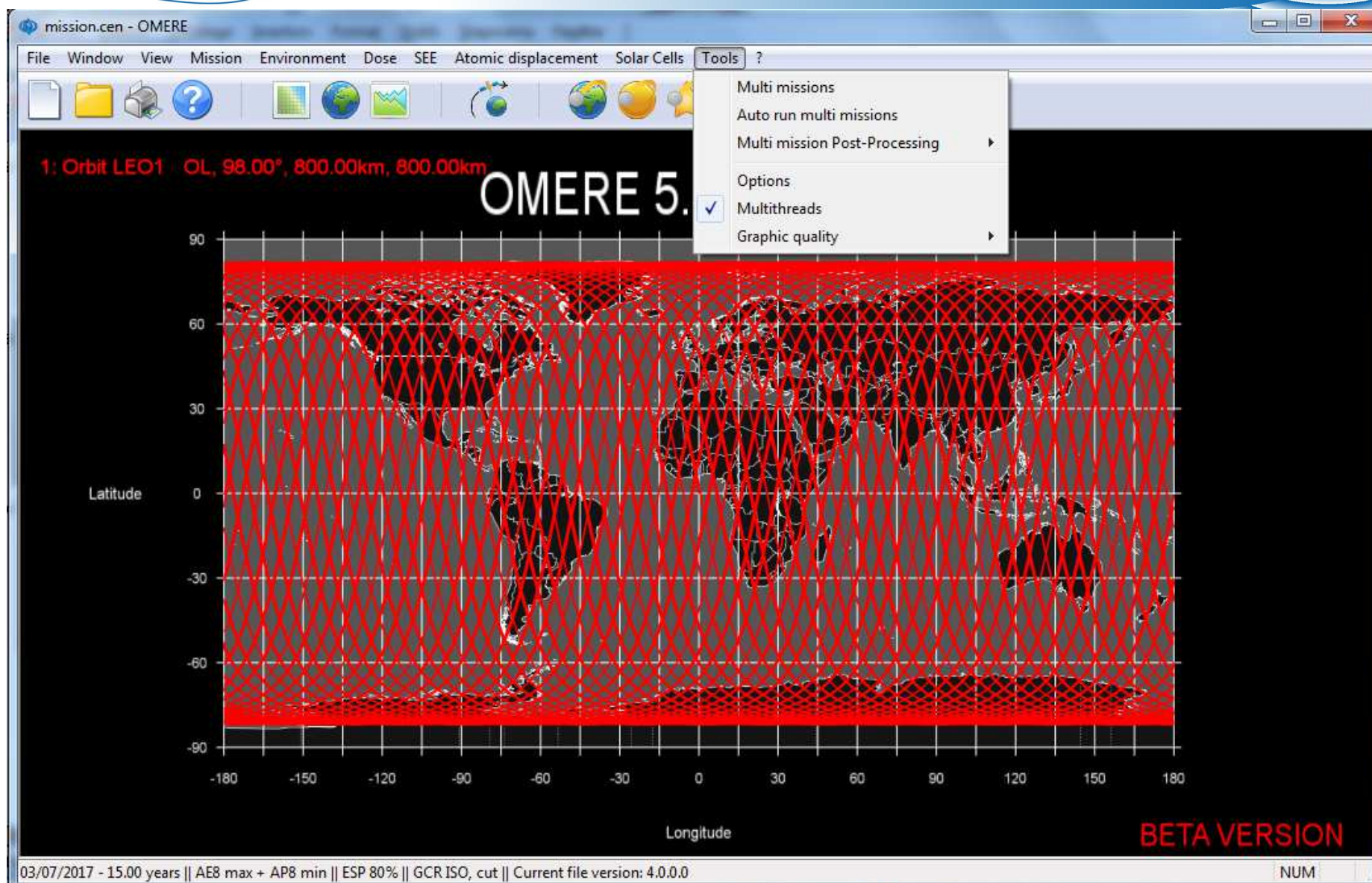
Altitude : 800.0 (km)
 Inclination : 98.0 (degree)
 Cosmic Rays : GCR ISO (SOL MIN)
 Trapped Protons : AP8 MIN (SOL MIN)
 Orbit Period : 100.87 (min)
 Mission Duration : 12.5 (years)
 SAA max location : -30° latitude/ -31° longitude

03/07/2017 - 15.00 years || AE8 max + AP8 min || ESP 80% || GC

OMERE







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Workshop OMERE
May 15 & 16, 2017 (CLS, Toulouse)

The Atmospheric & Space Environment CCT is pleased to announce the "Workshop OMERE" to be held at CLS (Toulouse) on 15 & 16 May 2017.

OMERE is a **freeware** dedicated to space environment and radiation effects on electronic

Crédits: NASA/GSFC/SDO
Cette journée d'animation avait pour objectif de faire un point sur les événements

The interface has been designed in collaboration with industrial partners in order to provide users with the fastest and most efficient way to perform radiation effect analysis. This software is freely available on the WEB since 2003.

<http://cct.cnes.fr/content/cct-env-workshop-omere>